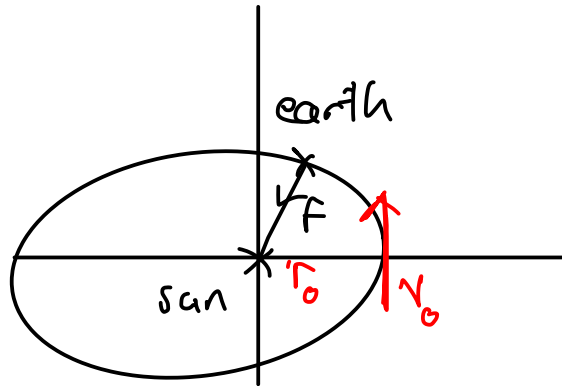


So far we've seen KII (equal areas in equal times) is equivalent to motion under central force.



Already shown:

$$\textcircled{1} \quad \underline{\ddot{x}} = \underbrace{(\ddot{r} - r(\dot{\theta})^2)}_{\text{radial acceleration}} \begin{bmatrix} \cos\theta \\ \sin\theta \end{bmatrix}$$

$$\textcircled{2} \quad r^2 \dot{\theta} = r_0 v_0, \text{ constant over all time.}$$

Can use this to eliminate $\dot{\theta}$

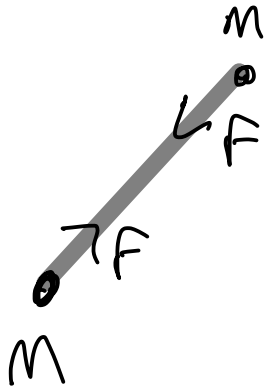
Need Newton's Law of Gravity

$$F = \frac{G M m}{r^2} \quad \text{force due to gravity}$$

G = universal gravitational constant

Two masses M, m

distance r apart



Now we "f=ma" to get

$$r^2 \dot{\theta} = r_0 v_0$$

$$-F = m (\ddot{r} - r (\dot{\theta})^2)$$

$$\therefore -\frac{GMm}{r^2} = m (\ddot{r} - r (\dot{\theta})^2)$$

$$\therefore GM = -r^2 \ddot{r} + \frac{r_0^2 v_0^2}{r}$$

A bit nasty-looking! Also want r in terms of θ not in terms of t .

Helpful to think what are we expecting?

Ellipse ... $r = \frac{r_0(1+e)}{1+e \cos \theta} \quad 0 \leq e < 1$

What diff. eq. could produce such a formula?

Suggest $u = \frac{1}{r}$ will work better

$$u = a \cos \theta + c$$

← This I do know a diff. eq. which has this as its solution.

$$\frac{d^2 u}{d\theta^2} + u = 0 \quad \text{has G.S.} \quad u = a \cos \theta + b \sin \theta$$

$$\frac{d^2 u}{d\theta^2} + u = c \quad \text{has G.S.} \quad u = \underbrace{a \cos \theta + b \sin \theta}_{\text{G.S. of associated homogeneous equation}} + \underbrace{c}_{\text{P.S.}}$$

So hoping if we set $u = \frac{1}{r}$, our diff. eq.

$$-r^2 \ddot{r} + \frac{r_0^2 v_0^2}{r} = GM$$

will transform into

$$\frac{d^2 u}{d\theta^2} + u = c \quad \dots$$

$$-r^2 \ddot{r} + \frac{r_0^2 v_0^2}{r} = GM$$

Let $u = \frac{1}{r}$

$$\frac{du}{dt} = -\frac{1}{r^2} \cdot \frac{dr}{dt}$$

$$\frac{du}{d\theta} \cdot \frac{d\theta}{dt} =$$

$$\therefore \frac{du}{d\theta} \cdot r^2 \dot{\theta} = -\dot{r}$$

$$\therefore \frac{du}{d\theta} \cdot r_0 v_0 = -\dot{r}$$

$$\frac{d}{dt} \left(\frac{du}{d\theta} \cdot \frac{d\theta}{dt} \cdot r_0 v_0 \right) = -\ddot{r}$$

$$\therefore \frac{d^2 u}{d\theta^2} \cdot r_0^2 v_0^2 = -r^2 \ddot{r}$$

$$\Rightarrow \frac{d^2 u}{d\theta^2} \cdot r_0^2 v_0^2 + \frac{r_0^2 v_0^2}{r} = GM$$

$$\therefore \frac{d^2 u}{d\theta^2} + u = \frac{GM}{r_0^2 v_0^2}$$

That's what we hoped!!

$$\therefore u = a \cos \theta + b \sin \theta + \frac{GM}{r_0^2 v_0^2}$$

$$\frac{du}{d\theta} = -a \sin \theta + b \cos \theta$$

When $\theta = 0$, $u = \frac{1}{r_0}$, $\frac{du}{d\theta} = 0$.

$$\therefore b = 0, \quad a + \frac{GM}{r_0^2 v_0^2} = \frac{1}{r_0}$$

$$\therefore a = \frac{GM}{r_0^2 v_0^2} \left[\frac{r_0 v_0^2}{GM} - 1 \right]$$

$$\therefore \frac{1}{r} = \frac{GM}{r_0^2 v_0^2} \left[\left(\frac{r_0 v_0^2}{GM} - 1 \right) \cos \theta + 1 \right]$$

$$\text{Let } e = \frac{r_0 v_0^2}{GM} - 1, \text{ so } r_0(e+1) = \frac{r_0^2 v_0^2}{GM}$$

$$r = \frac{r_0^2 v_0^2}{GM} \cdot \frac{1}{1 + e \cos \theta}$$

$$\therefore r = \frac{r_0(1+e)}{1+e \cos \theta} \quad \text{where } e = \frac{r_0 v_0^2}{GM} - 1 = \left(\frac{v_0}{v_{\text{crit}}} \right)^2 - 1$$

$$v_{\text{crit}} = \sqrt{\frac{GM}{r_0}}$$

General equation
for a conic section
eccentricity e

This is our final solution:

$$v_{\text{crit}} = \sqrt{\frac{GM}{r_0}}, \quad e = \left(\frac{v_0}{v_{\text{crit}}} \right)^2 - 1 \quad \text{Then} \quad r = \frac{r_0(1+e)}{1+e \cos \theta}$$

$$v_{\text{crit}} = \sqrt{\frac{GM}{r_0}}, \quad e = \left(\frac{v_0}{v_{\text{crit}}} \right)^2 - 1 \quad \text{Then} \quad r = \frac{r_0(1+e)}{1+e \cos \theta}$$

Think about e ...

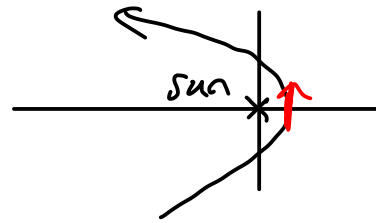
$$v_0 > \sqrt{2} v_{\text{crit}} \Leftrightarrow e > 1$$

$$v_0 = \sqrt{2} v_{\text{crit}} \Leftrightarrow e = 1$$

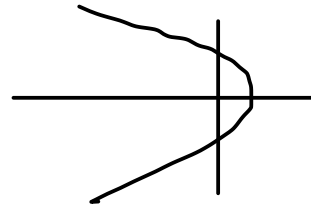
$$v_{\text{crit}} < v_0 < \sqrt{2} v_{\text{crit}} \Leftrightarrow 0 < e < 1$$

$$v_0 = v_{\text{crit}} \Leftrightarrow e = 0$$

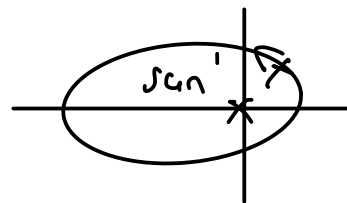
$$0 < v_0 < v_{\text{crit}} \Leftrightarrow -1 < e < 0$$



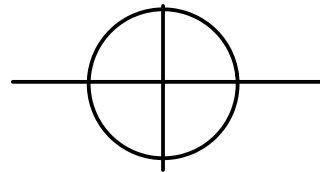
hyperbola, planet escapes sun's gravity to infinity.



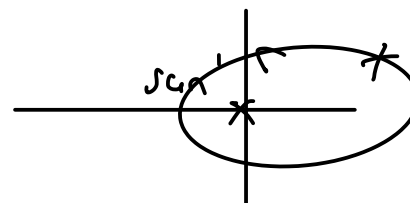
parabola, still escapes



Ellipse ✓



Circle



Another ellipse, other focus!

We've proved KI from NL of G.

Finally derive K_{III} ... period T for one full orbit.

Assume $0 \leq e < 1$

From K_{II} , know area swept out in time t is $\frac{1}{2} r_0 v_0 t$

Period T is time when area is area of ellipse

$$A = \frac{1}{2} \int_0^{\theta} r^2 \cdot d\theta = \frac{1}{2} \int_0^t \underbrace{r^2 \dot{\theta}}_{r_0 v_0} dt = \frac{1}{2} r_0 v_0 t$$

So

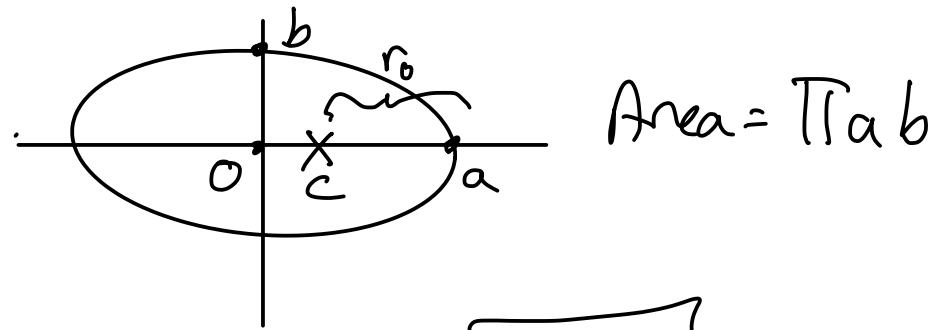
$$\frac{1}{2} r_0 v_0 T = \pi a^2 \sqrt{1-e^2}$$

$v_0 = v_{crit} \sqrt{1+e} = \frac{\sqrt{GM}}{\sqrt{r_0}} \sqrt{1+e}$

$$\therefore \frac{\sqrt{r_0} \sqrt{GM} \sqrt{1+e}}{2 \sqrt{r_0}} T = \pi a^2 \sqrt{1+e} \sqrt{1-e^2}$$

$$\therefore \frac{\sqrt{a} \sqrt{1+e} \sqrt{GM}}{2} T = \pi a^2 \sqrt{1+e}$$

$$T = \frac{2\pi a^{3/2}}{\sqrt{GM}}$$



$$c = ae$$

$$e = \sqrt{1 - \frac{b^2}{a^2}}$$

$$r_0 = a - c$$

$$= a - ae$$

$$\therefore r_0 = a(1-e)$$

$$b = a \sqrt{1-e^2}$$

So :

$$T = \frac{2\pi a^{3/2}}{\sqrt{GM}}$$

KIII

$$\frac{a^3}{T^2} = \frac{GM}{4\pi^2}$$

Constant independent of r_0, v_0 — the planet!

Shows $\frac{a^3}{T^2}$ is constant for all planets.

You can work out mass of sun!!!